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Group Report

1964-57

Haystack Pointing System: Interpolation R. Teoste

28 October 1964

Prepared under Electronic Systems Division Contract AF 19 (628)-500 by

# Lincoln Laboratory

MASSACHUSETTS INSTITUTE OF TECHNOLOGY

Lexington, Massachusetts



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# MASSACHUSETTS INSTITUTE OF TECHNOLOGY LINCOLN LABORATORY

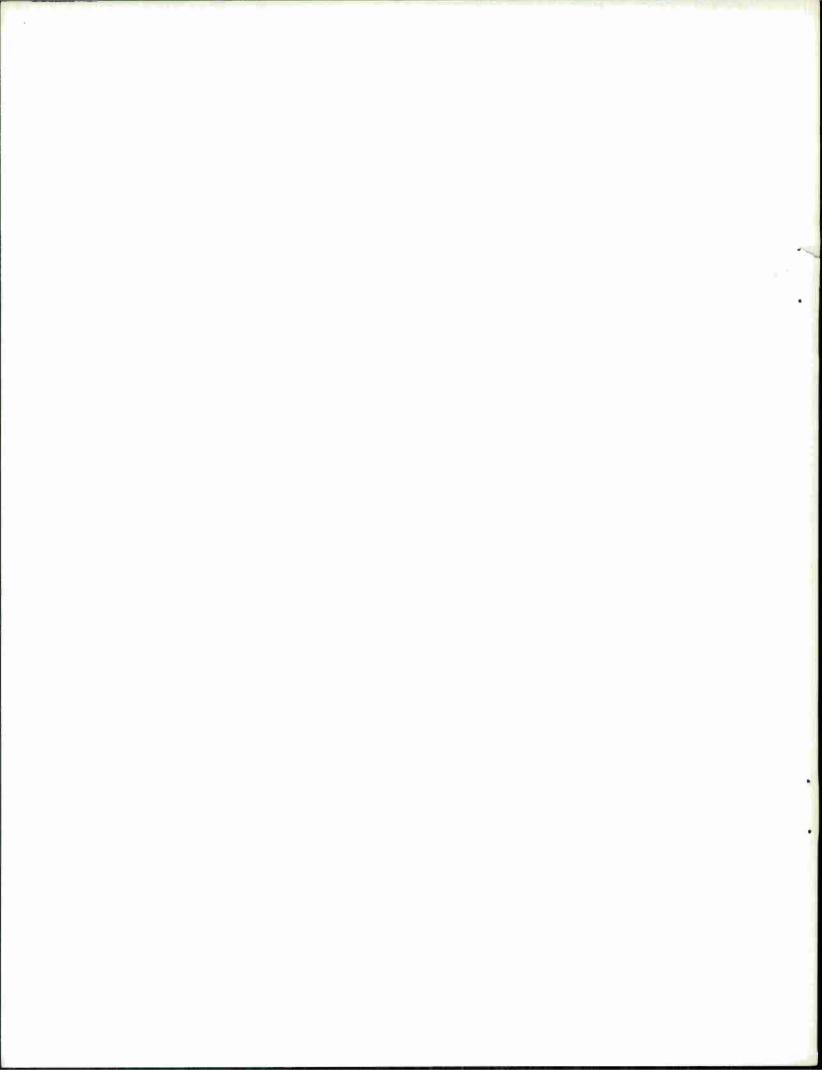
HAYSTACK POINTING SYSTEM: INTERPOLATION

R. TEOSTE

Group 62

GROUP REPORT 1964-57

28 OCTOBER 1964



## ABSTRACT

The Haystack antenna is pointed by means of a Univac 490 Computer. In the Pointing System program is incorporated an interpolation routine to provide pointing information 250 times a second. This report describes the interpolation routine.

Accepted for the Air Force Stanley J. Wisniewski Lt Colonel, USAF Chief, Lincoln Laboratory Office

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## HAYSTACK POINTING SYSTEM: INTERPOLATION

## INTRODUCTION

The Interpolation subroutine of the Haystack Pointing System is the last program to handle pointing information before it leaves the computer. The subroutine operates in the two second cycle of the Pointing System without any operator intervention. It interpolates for the required number of data points and converts the pointing information into proper format for the external equipment.

The Interpolation program accepts azimuth and elevation angles computed at two second intervals and interpolates 500 values of each at 4 millisecond intervals. The interpolation is handled so that dynamic servo error corrections can be added to the azimuth and elevation angles. The program computes doppler from range rate, makes a time correction for one-way transit time, and similarly interpolates to get 500 doppler values for every two second interval. Also the average range is computed for each two second interval.

## PROGRAM INPUTS

All program inputs are obtained from core storage. The following inputs are required:

W(ACQAZIM)	B27 azimuth angle for $T_0$ -2 sec (Revolutions)
W(ACQAZIM+1)	B27 azimuth angle for $T_0$ (Revolutions)
W(ACQAZIM+2)	B27 azimuth angle for $T_0$ +2 sec (Revolutions)
W(ACQAZIM+3)	B27 azimuth angle for T +4 sec (Revolutions)
W(ACQELEV)	B27 elevation angle for $T_0$ -2 sec (Revolutions)
W(ACQELEV+1)	B27 elevation angle for $T_0$ (Revolutions)
W(ACQELEV+2)	B27 elevation angle for $T_0+2$ sec (Revolutions)
W(ACQELEV+3)	B27 elevation angle for $T_0 + 4$ sec (Revolutions)
W(CRANGE)	BO range for $T_0 + 4$ sec (.2 µsec of two-way range)
W(RANGEDOT)	B24 range rate for To+4 sec (nm/sec)
L(AZIMADD)	First location of azimuth 500 word buffer
L(ELEVADD)	First location of elevation 500 word buffer
L(RANGEADD)	Location of average range

L(DOPPADD) First location of doppler 500 word buffer

L(RADARMODE) Code for one- or two-way doppler

+ for two-way doppler

- for one-way doppler

## PROGRAM OUTPUTS

The program outputs and the word formats are shown in TABLE I.

TABLE I PROGRAM OUTPUTS

	Azimuth	Elevation	Doppler	Range
Bit Positions	0-19	0-18	0-20	0-24
Binary Point	<b>B</b> 19	Bl9	во	BO
Units	Revolutions	Revolutions	CPS	.2 µsec of two-way range
Bias	0 = North	0 = Horizon	0 = -750,000	0 = 0
Range	0999998	0999998	75-+.75MC	0-6.7 sec
Comments	A one in Bit Position 19 indicates overlap zone	Angles below horizon are represented by 3600-(angle)		
No. of Words	500	500	500	1
First Word of Buffer Specified By	L(AZIMADD)	L(ELEVADD)	L(DOPPADD)	L(RANGEADD)

## INTERNAL OPERATION

## Azimuth Interpolation

Let us designate four points of the desired antenna azimuth angle by  $F_{-1}$ ,  $F_0$ ,  $F_1$ , and  $F_2$  for  $T_0$  - 2 sec,  $T_0$ ,  $T_0$  + 2 sec and  $T_0$  + 4 sec, respectively. The values of azimuth angles between  $T_0$  and  $T_0$  + 2 sec are to be interpolated. The interpolation is done in two parts: first, 25 equally-spaced points are interpolated between  $F_0$  and  $F_1$  by a four-point formula, and

then 9 points will be inserted between these interpolated points by linear interpolation.

The four-point formula, which will also insert a time shift and perform dynamic compensation, is of the form

$$F(y_k) = \sum_{i=-1}^{2} a_i (y_k) F_i , \qquad (1)$$

where

$$\begin{aligned} \mathbf{a}_{-1}(\mathbf{y}_{k}) &= \mathbf{c}_{-13}(\mathbf{y}_{k}+\Delta)^{3} + \mathbf{c}_{-12}(\mathbf{y}_{k}+\Delta)^{2} + \mathbf{c}_{-11}(\mathbf{y}_{k}+\Delta) + \mathbf{c}_{-10} \\ \mathbf{a}_{0}(\mathbf{y}_{k}) &= \mathbf{c}_{03}(\mathbf{y}_{k}+\Delta)^{3} + \mathbf{c}_{02}(\mathbf{y}_{k}+\Delta)^{2} + \mathbf{c}_{01}(\mathbf{y}_{k}+\Delta) + \mathbf{c}_{00} \\ \mathbf{a}_{1}(\mathbf{y}_{k}) &= \mathbf{c}_{13}(\mathbf{y}_{k}+\Delta)^{3} + \mathbf{c}_{12}(\mathbf{y}_{k}+\Delta)^{2} + \mathbf{c}_{11}(\mathbf{y}_{k}+\Delta) + \mathbf{c}_{10} \\ \mathbf{a}_{2}(\mathbf{y}_{k}) &= \mathbf{c}_{23}(\mathbf{y}_{k}+\Delta)^{3} + \mathbf{c}_{22}(\mathbf{y}_{k}+\Delta)^{2} + \mathbf{c}_{21}(\mathbf{y}_{k}+\Delta) + \mathbf{c}_{20} \end{aligned}$$

and

C; are the interpolation coefficients

h is the interval between computed points (2 seconds)  $y_k$  is the normalized time,  $y_k = \frac{t_k - T_0}{h}$ 

is the normalized desired time shift  $\triangle = \frac{\text{time}}{h}$ 

The remaining points are computed by linear interpolation

$$F_{\ell} = \frac{\ell}{10} \left[ F(y_{k+1}) - F(y_k) \right] + F(y_k) , \qquad (3)$$

where  $\ell$  is an integer which ranges from zero to nine, and F, is the value of the desired function.

The errors due to these interpolation formulae are negligible as compared to the system error for typical expected target trajectories.

The basic quantities that do the work are C of Eq. (2). The values of the coefficients can be computed so that some of the dynamic servo error is eliminated. If no servo compensation is desired, the coefficients are

derived from a third-difference Bessel's interpolation formula.\* The time delay  $\Delta$  is added to take out the delays of the interface equipment and asynchronism of the computer and encoder systems.

TABLE II shows the interpolation coefficients and a correction term to each. The servo dynamic error is expected to be a linear combination of the antenna angular rates. The correction term of each coefficient in TABLE II adds a correction to the interpolated values of the form

$$E = K_0 F + K_1 \frac{dF}{dt} + K_2 \frac{d^2 F}{dt^2} + K_3 \frac{d^3 F}{dt^3} , \qquad (4)$$

where

E is in degrees

F is in degrees

K is dimensionless

 $K_1$  is in seconds

K is in seconds<sup>2</sup>

 $K_3$  is in seconds<sup>3</sup>.

While any correction that is of the form of Eq. (4) can be added, the values of  $K_i$  are related to the servo error constants. The values of the constants will have to be measured experimentally.

## ELEVATION INTERPOLATION

The elevation pointing angles are interpolated by the same method as the azimuth angles. Since the elevation servo response is expected to differ from the azimuth servo response, the values of the correction terms in the interpolation coefficients will be different.

RANGE

The average range for the interval between  $\mathbf{T}_{0}$  and  $\mathbf{T}_{0}$ + 2 sec is computed by

$$R_{\text{avg}} = \frac{1}{2} \left( R_0 + R_1 \right) \tag{5}$$

<sup>\*</sup>D. R. Hartree, "Numerical Analysis," (Oxford University Press, London, 1955), p. 68.

<sup>&</sup>lt;sup>+</sup>J. G. Truxal, "Automatic Feedback Control System Synthesis," (McGraw-Hill, New York, 1955), p. 82.

TABLE II
INTERPOLATION COEFFICIENTS

	Interpolation Coefficient	CorrectionTerm
C <sub>-13</sub>	- 1/6	- <sup>K</sup> 0 6
C_12	1/2	$\frac{K_0}{2} - \frac{K_1}{4}$
C_11	<b>-</b> 1/3	$-\frac{K_0}{3}+\frac{K_1}{2}-\frac{K_2}{4}$
C_10	0	$-\frac{K_1}{6} + \frac{K_2}{4} - \frac{K_3}{8}$
C <sub>03</sub>	1/2	K <sub>O</sub> 2
C <sub>02</sub>	- 1	$- K_0 + \frac{3K_1}{4}$
C <sub>Ol</sub>	- 1/2	$-\frac{K_0}{2}-K_1+\frac{3K_2}{4}$
C <sub>00</sub>	1	$K_0 - \frac{K_1}{4} - \frac{K_2}{2} + \frac{3K_3}{8}$
C <sub>13</sub>	- 1/2	- <sup>K</sup> O 2
C <sub>12</sub>	1/2	$\frac{K_0}{2} - \frac{3K_1}{4}$
$\mathbf{C}_{\texttt{ll}}$	ı	$K_0 + \frac{K_1}{2} - \frac{3K_2}{4}$
C <sub>lo</sub>	0	$\frac{K_1}{2} + \frac{K_2}{4} - \frac{3K_3}{8}$
C <sub>23</sub>	1/6	K <sub>O</sub> 6 K <sub>1</sub> 4
C <sub>22</sub>	0	$\frac{K_{\underline{1}}}{\underline{4}}$
C <sub>21</sub>	- 1/6	$-\frac{K_0}{6}+\frac{K_2}{4}$
<sup>C</sup> 20	0	$-\frac{K_1}{12}+\frac{K_3}{8}$

Where R is the computed range at  $T_0$  and

 $R_1$  is the computed range at  $T_0$  + 2 sec.

## DOPPLER INTERPOLATION

At main bang time, the radar receiver will request a doppler frequency. The frequency that it expects to get should correspond to the doppler shift due to the instantaneous range rate at the time the radar pulse hits the target. Hence, the doppler shift should be computed for one-way transit time later than when it is requested. Since the range is a slowly-varying function, and since the doppler shift is approximately a linear function of time, we can compute the one-way transit time and an average doppler rate and add their product to the doppler frequency at main bang time.

The one-way doppler shift is approximated by

$$D = -\frac{R}{C} \left(1 - \frac{R}{C}\right) \text{ fo} \qquad , \tag{6}$$

where

D is the doppler shift (cps)

C is the speed of light (nm/sec)

fo is the transmitted frequency (cps)

R is the range rate (nm/sec)

When the one-way transit time is added into the calculations, Eq. (6) becomes

$$D_{\text{one-way}} = -\frac{fo}{C} \left[ \dot{R} \left( 1 - \frac{\dot{R}}{C} \right) + \frac{\dot{R}}{C} \frac{\dot{dR}}{dt} \right]$$
 (7)

<sup>\*</sup>F. W. Sears and M. W. Zemansky, "College Physics," (Addison-Wesley, Cambridge, Massachusetts, 1952),p. 403.

The two-way doppler is twice Eq. (7), or

$$D_{\text{two-way}} = -\frac{2fo}{C} \left[ R \left( 1 - \frac{R}{C} \right) + \frac{R}{C} \right] \frac{dR}{dt} \right] . \tag{8}$$

Four values of range rate are converted to doppler by Eq. (7) or (8) depending whether one-or two-way doppler was requested, and 250 points per second are interpolated as is done in the azimuth - elevation calculations by computing 25 points per second with the four-point formula and linearly interpolating for the remaining points. In the doppler interpolation, no dynamic compensation is required; thus, the coefficients C<sub>ij</sub> are given by the first column of TABLE II.

The interpolation formulae contribute no measurable errors in doppler shift.

### PROGRAM DETAILS

The interpolations can be speeded up by working with the differences of the function values. This suggests the following transformation:

$$F(y) = a_{-1}(y)F_{-1} + a_{0}(y)F_{0} + a_{1}(y)F_{1} + a_{2}(y)F_{2}$$

$$= a_{-1}(y)(F_{-1} - F_{0}) + a_{1}(y)(F_{1} - F_{0}) + a_{2}(F_{2} - F_{0}) + F_{0}.$$
(9)

Here only three multiplications are required instead of the four in the original form.

The program for computing this interpolation consists of two main parts - an initialization portion and an interpolation portion. The initialization portion is used while the Pointing System is being initialized and the interpolation portion of used in every two second cycle. Since the interpolation portion of the program has to run very fast, a lot of computations were done in the initialization portion of the program.

Figure 1 shows the flow chart of the initialization portion of the program. This portion of the program contains a subroutine (INITSUB) which computes the 51 values of  $a_i(y)$ . The subroutine requires that the starting

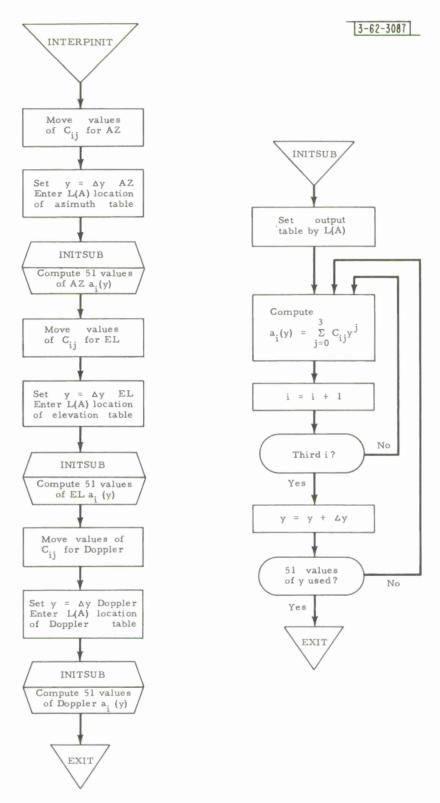


Fig. 1. Flow chart of the initialization portion of interpolation subroutine.

value of y be stored in core storage and the first location of the table for  $a_i(y)$  be in the lower half of the A register. The order of the coefficients  $a_i(y)$  which the subroutine computes is:  $a_{-1}(y_1)$ ,  $a_1(y_1)$ ,  $a_2(y_1)$ ,  $a_{-1}(y_2)$ ,  $a_1(y_2)$ ,  $a_2(y_2)$ , etc. The control program which uses this subroutine simply moves the values of  $C_{ij}$  into the location which the subroutine has set aside for them and sets the initial value of y to the timing delay associated with the particular variable. The control program does this for azimuth, elevation and doppler.

Since the initialization portion of the program is run only once during a pass, no effort was made to make the program fast. The required time for interpolation program initialization is a fraction of a second.

The interpolation portion of the program is flow charted in Fig. 2. It also contains a subroutine. The subroutine (INTSUB) interpolates 51 values of the function and fills in 9 points between these values by linear interpolation. The subroutine requires the values  $(F_{-1} - F_0)$ ,  $(F_1 - F_0)$ ,  $(F_2 - F_0)$ , and  $F_0$ . In addition, the location of the table of  $a_i(y)$  and the location of the output buffer must be specified by index registers B4 and B5, respectively. The program which controls this subroutine (INTERP), supplies the values of  $(F_{-1} - F_0)$ ,  $(F_1 - F_0)$ ,  $(F_2 - F_0)$ , and  $F_0$  to the subroutine and loads index registers B4 and B5 with proper table locations. The azimuth and elevation angles are compared to maximum and minimum limits and sent to the INTERP subroutine which computes the required pointing angles. The doppler values are first computed from range rate, and then the subroutine is used.

The interpolation portion of the interpolation routine was made as fast as possible. Some portions were coded several times in order to save time which is required for looping instructions. The running time of the interpolation portion of the program was measured to be 140 milliseconds for a particular test run. The timing will vary slightly for different input functions.

The complete listing of the interpolation program is given in the Appendix.

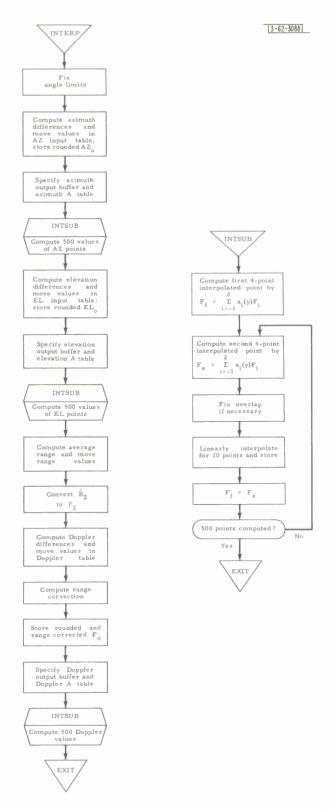


Fig. 2. Flow chart of the interpolation portion of subroutine.

Some of the variables one might want to change are:

W(DELAZ)	Azimuth timing delay	B27 in fractions of 2 sec interval (Present value in program is 1.806 m sec)
W(DELEL)	Elevation timing delay	B27 in fractions of 2 sec interval (Present value in program is 1.812 m sec)
W(DELDOP)	Doppler timing delay	B27 in fractions of 2 sec interval (Present value in program is 0)
W(CAZ)	Azimuth interpolation coefficients	B27 in the order C <sub>20</sub> , C <sub>21</sub> , C <sub>22</sub> , C <sub>23</sub> , C <sub>10</sub> , (Present values in program are the ones given in TABLE II column labeled interpolation coefficient)
W(CEL)	Elevation interpolation coefficients	B27 in the order C <sub>20</sub> , C <sub>21</sub> , C <sub>22</sub> , C <sub>23</sub> , C <sub>10</sub> , (Present values in program are the ones given in TABLE II column labeled interpolation coefficient)
W(CDOP)	Doppler interpolation coefficients	B27 in the order C <sub>20</sub> , C <sub>21</sub> , C <sub>22</sub> , C <sub>23</sub> , C <sub>10</sub> , (Present values in program are the ones given in TABLE II column labeled interpolation coefficient)
W(ELMIN)	Minimum allowable elevation	B27 in revolutions (Present value in program is -1.8°)
W(EIMAX)	Maximum allowable elevation	B27 in revolutions (Present value in program is 90°)
W(AZMIN)	Minimum allowable azimuth	B27 in revolutions (Present value in program is -115.2°)
W(AZMAX)	Maximum allowable azimuth	B27 in revolutions (Present value in program is 475.2°)

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63334

## SPURT OUTPUT NO. 211

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	INTER	TEOSTE+06/19/	54		
LABEL	LOC	LABEL	LOC	LABEL	LOC
MINREG	63152	NMPERAU	63340	POLE	63324
PLANP	63434	PRLOG	63423	R	01460
R1	00151	R2	00153	R3	00157
R4	00164	R5	01356	RA	63002
RADOT	63007	RADARMODE	63312	RADIOMETER	63102
RADIUS	63006	RADIUSDOT	63011	RANGE	63052
RANGEOUT	70777	RANGEADD	63445	RANGEDOT	63062
RANGEUNIT	01334	RDOT	01463	RDMTR	63430
RDXXX	63433	RECORDSIZE	63112	RECAZIM	67000
RECC	00057	RECELEV	70000	RECFILE	63212
RECRD	63415	REVOLUTION	01333	SAZIM	
SCELTIME	63134	SDEC	63005	SECOND	00202
SECONDS	63140	SELEV	63056	SIDERTIME	63012
SINORIENT	63064	SINAZEL	63066	SKIP	63331
SRA	63004	SRADTIME	63136	STARTAZ	01130
STARTEL	01143	SYSENTRIES	77600	SYSNAMES	77700
SYSTAT1	63313	SYSTAT2	63314	SYSTATD	63315
TENTH	01457	TEST1	01303	TEST2	01311
TEST3	01315	TEST5	01322	TIMECORR	63107
TIMEMODE	63103	TIMEP	63435	TRUERANGE	63063
VELOFLIGHT	63335	VIZDEC1	63014	VIZDEC2	63016
VIZRA1	63013	VIZRA2	63015	WFORD	
WFADD	63450	WFFREQ	63333	Y	00060
YEARMONTH	63147	YRTRAN	63327	ZRTRAN	63330

END OF LISTING

## ..... SPURT OUTPUT NO. 212 .....

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	INTER	TEOSTE + 06/19/6	54		
LABEL	LOC	LABEL	LOC	LABEL	LOC
INTERY	00000	INTERPINIT	00002	A\$\$\$\$\$1111	00004
A\$\$\$\$\$1112	00010	A\$\$\$\$\$1113	00020	A\$\$\$\$\$1114	00024
A\$\$\$\$\$1115	00034	A\$\$\$\$\$1116	00040	UKS	00056
RECC	00057	Y	00060	CONVERSION	00061
DELAZ	00062	DELEL	00063	DEL DOP	00064
CONST	00065	CEL	00066	CAZ	00106
CDOP	00126	INI TSUB	00146	R1	00151
R2	00153	R3	00157	R4	00164
DELY	00177	JUNK	00200	FIRST	00201
SECOND	00202	F	00210	AAZ	00214
AEL	00445	ADOP	00676	INTERP	01127
STARTAZ	01130	ENDAZ	01135	ENDAZBOUNC	01142
STARTEL	01143	ENDEL	01150	TEST1	01303
BOUNCE	01310	TEST2	01311	TEST3	01315
TEST5	01322	LAP	01326	ELMIN	01327
ELMAX	01330	AZMIN	01331	AZMAX	01332
REVOLUTION	01333	RANGEUNIT	01334	BIAS	01335
INTSUBR	01336	R5	01356	TENTH	01457
R	01460	RDO T	01463	I D1 CELCOR	63000
I D2CELCOR	63001	RA	63002	DEC	63003
SRA	63004	SDEC	63005	RADIUS	63006
RADOT	63007	DECDOT	63010	RADIUSDOT	63011
SIDERTIME	63012	VIZRA1	63013	VIZDEC1	63014
VIZRA2	63015	VIZDEC2	63016	ID1RADCOR	63050
I D 2R AD COR	63051	RANGE	63052	AZIM	63053
ELEV	63054	SAZIM	63055	SELEV	63056
CRANGE	63057	CAZIM	63060	CELEV	63061
RANGEDOT	63062	TRUERANGE	63063	SINORIENT	63064
COSURIENT	63065	SINAZEL	63066	COSAZEL	63070
AC QAZIM EL	63071 63075	AZ EDA :: EST ZE	63071 63101	ACQELEV	63075
TIMEMODE	63103	FRAMESI ZE		RADIOMETER	63105
ASTRODEC	63106	FIRSTELEV TIMECORR	63104 63107	ASTRORA KYBRDLEVEL	63110
TTYSTATUS	63111	RECORDSIZE	63112	CELBODY	63113
ID1TIME	63130	IDZTIME	63131	TRUETIME	63132
CELTIME	63133	SCELTIME	63134	CONVERTIME	63135
SRADTIME	63136	HOURMINUTE	63137	SECONDS	63140
DSECONDS	63141	ACTUALTIME	63142	ESTSHIFTED	63143
GMTSHIFTED	63144	GMTMODU24	63145	BLASTOFF	63146
YEARMONTH	63147	DAY	63150	HOURREG	63151
MINREG	63152	FIRSTHRU	63153	DUMSECTIG	63154
I D1RECRD	63210	IDZRECRD	63211	RECFILE	63212
ID1SYSPAR	63310	ID2 SYSP AR	63311	RADARMODE	63312
SYSTAT1	63313	SYSTAT2	63314	SYSTATO	63315
DELTATEE	63316	FREQUENCY	63317	LONGITUDE	63320
GEODETLAT	63321	GEOCENLAT	63322	EQUATOR	63323
POLE	63324	AZIMOVER	63325	HEIGHT	63326
YRTRAN	63327	ZRTRAN	63330	SKIP	63331
AUCONVER	63332	WFFREQ	63333	MAINSWITCH	63334
VELOFLIGHT	63335	LSPERAU	63336	FLATTENING	63337
NMPERAU	63340	AUPEREQUAT	63341	KMPERNM	63342

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#### SPURT OUTPUT NO. 212 INTER TEOSTE+06/19/64 LABEL LABEL LOC LOC LABEL LOC ID2ENTPNT EXPNAME 63350 ID1ENTP NT 63410 63411 MCPGM 63412 INTER 63413 COCON 63414 63415 63417 RECRD ADSCN 63416 AESCN CORCT 63420 DYDMP 63421 63422 CHCOR PRLOG 63423 CELCOMPGM 63424 DATANALYZE 63425 INTERCOM 63426 ACQUI 63427 RDMTR 63430 CHPAR 63431 WFORD 63432 RDXXX 63433 PLANP 63434 TIMEP 63435 ID1RADIO 63440 ID2RADIO 63441 AZIMADD 63442 ELEVADD 63443 DOPPADD 63444 RANGEADD 63445 INAZIMADD 63446 INELEVADD 63447 WFADD 63450 ID3RADIO 63776 ID4RADIO 63777 AZIMOUT 64000 ID5RADIO 64776 In6RADIO 64777 **ELEVOUT** 65000 ID7RADIO 65776 I D8RADIU 65777 66000 DOPPOUT ID9RADIO 66776 IDIORADIO 67000 66777 RECAZIM ID11RADIO 67776 ID12RADIO 67777 70000 RECELEV ID13RADIO 70775 In14RADIO 70776 RANGEOUT 70777 MCPFILLER 71000 ID15RADIO 71776 72000 ID16RADIO 71777 INTERAZIM ID17RADIO 72776 ID18RADIO 72777 INTERELEV 73000 ID19RADIO 73776 73777 74000 ID2ORADIO INTERDOPP In21RADIO 74776 ID22RADIO 74777 AZIMIN 75000 ID23RADIO 75776 ID24RADIO 75777 ELEVIN 76000 ID25RADIO 76775 INTERRANGE 76777 ID26RADIO 76776

77577

77677

SYSENTRIES

SYSNAMES

77600

77700

ID2 SYSENT

ID2SYSNAM

END OF LISTING

ID1SYSENT

ID1SYSNAM

77576

77676

SPURT OUTPUT NO. 210

INTER TEOSTE+06/19/64

NO.	OF	INST	RUCTIONS	01467	
	00	000	THRU	00200	
	00	1214	THRU	00214	
	00	445	THRU	00445	
	00	1676	THRU	00676	
	01	127	THRU	01460	
	01	463	THRU	01463	
	0.1	466	THRU	01466	

## SPURT OUTPUT NO. 210 INTER TEOSTE • 06/19/64

		THIEK IEU21E	*U0/19/04	
CARDS	L1 ID LABEL T	A STATEMENT	LOC F JKB Y	NOTES
	00000 1117-0	A STATEMENT  PROGRAM TEDSTE*06/19/64  U-TAG INTERP*INTERPINIT  FD 1*INTER ENTRY MOVE 8D*CAZ*JUNK  MOVE 4*CAZ*12D*JUNK*8D		
•	OOOOO INTER	PRUGRAM TEUSTE*UD/19/64	00000 01127 0000-	
•	OCCUPATION OF THE PROPERTY OF	U-TAG INTERPAINTERPINIT	00000 01127 00002	
	00002	FU I+INIEK	00001 16233 11227	
•	UUUU3 INTERPINIT	ENTRY	00002 61000 00000	
•	00004	MUVE 80*CAZ*JUNK	00003 12700 00007	MUVE VALUES UP AAZ
			00004 10037 00106	
			00005 14037 00200	
	00005	MOVE 4-CA7+12D- HINKARD	00000 12700 00004	
•	00005	MUVE 4+CAZ+IZU+JUNK+8U	00007 12700 00003	
			00010 10037 00122 00011 14037 00210	
		MOVE 4+CAZ+12D+JUNK+8D	00011 14037 00210	
	00006	MOVE 1aDELAZAV	00012 72700 00010	DELAZ FOLIALS V
•	00000	HOVE INDELACT	00013 10030 00062	DELAZ EGONES 1
	00007	ENT AMAA7	00014 14030 00000	OUTDUT TABLE IN AA7
•	00010	MOVE 1*DELAZ*Y  ENT A*AAZ RJP INITSUB MOVE 8D*CEL*JUNK	00015 11000 00214	COMPLITE AT I MITH TABLE
	00011	MOVE 80+CEL+JUNK	00017 12700 00007	MOVE VALUES OF CEL
•	00011	NOVE OD OEE ON	00011 12100 00001	HOVE VALUES OF CLE
			00021 14037 00200	
			00022 72700 00020	
	00012	MOVE 4+CEL+12D+JUNK+8D	00023 12700 00003	
			00024 10037 00102	
			00035 14037 00310	
			00026 72700 00024	
	00013	MOVE 1+DELEL+Y	00027 10030 00063	DELEL EQUALS Y
			00026 72700 00024 00027 10030 00063 00030 14030 00060 00031 11000 00445 00032 65000 00146 00033 12700 00007 00034 10037 00126	
	OC014	ENT A+AEL RJP INITSUB	00031 11000 00445	OUTPUT TABLE IN AEL
		RJP INITSUB	00032 65000 00146	COMPUTE ELEVATION TABLE
	00016	MOVE 8D*CDOP*JUNK	00033 12700 00007	MOVE VALUES OF COOP
			00034 10037 00126	
		MOVE 4*CDOP+12D*JUNK+8D	00035 14037 00200	
			00036 72700 00034	
•	00017	MOVE 4*CDOP+12D*JUNK+8D	00037 12700 00003	
			00040 10037 00142	
			00041 14037 00210	
		water a second of	00042 72700 00040	
•	00020	MOVE 1+DELDOP+Y	00043 10030 00064	DELDOP = Y
	00001	ENT. A-AD-	00044 14030 00060	OUTDUT TABLE IN ADDR
•	00021	ENI A*AUUP	00045 11000 00676	OUTPUT TABLE IN ADUP
•	00022	KJP INIISUB	00047 10030 00045	COMPOSE DOPPLEK TABLE
•	00023 00024	ENT AMIX(DADADMODE) AADOS	00050 11650 62212	CODE IS BUILS ECR 2-MAY DORDI ER
•	00024	MOVE 1*DELDOP*Y  ENT A*ADOP  RJP INITSUB  ENT Q*W(CONST)  ENT A*LX(RADARMODE)*APOS	00030 11630 63312	CODE 13 PLUS PCK 2-WAY DUPPLEK
	00025	RSH Q+1	00051 01000 00001	
	00026	MUL W(FREQUENCY)	00052 22030 63317	
	00027	RSH AQ#28D	00053 03000 00034	810
	00030	RSH Q*1 MUL W(FREQUENCY) RSH AQ*28D STR Q*W(CONVERSION)	00054 14030 00061	CONVERSION CONSTANT 2F/C OR F/
				С
•	00031	EXIT	00055 61010 00002	
•	00032 UKS	010000000	00056 01000 00000	DEC 1824
•	00033 RECC 00034 Y	0000014757	00057 00000 14757	DEC .000006182830
	UUU34 Y	U	00060 00000 00000	

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				*					
CARDS	L1 ID	LABEL	TA	STATEMENT	LOC	F JKB	Υ	NOTES	
•	00035	CONVERSION DELAZ		RESERVE 1 0000354556	00061 00062		00000 54556	DEC IMUTH TIM	.000903827 AZ
•	00037	DELEL		0000355401	00063	00003	55401	DEC	.000906827 EL
•	00040	DELDOP		000000000	00064	00000	00000	DEC ING DELAY	0.0827 DOPPLER TIM
٠	00041	CONST		1427243740	00065	14272	43740	DEC	12.364378824 2/C E-
•	00042	CEL		0000000000	00066	00000	00000	DEC	0.0827 EVELATION I
•	00043			7652525252	00067	76525	25252	DEC	ON COEFFIC 16666667B27
•	00044			0000000000	00070	00000	00000	DEC	0827
•	00045			0125252525	00071	01252	52525	DEC	-16666667827
•	00046			0000000000	00072	00000	00000	DEC	0827
•	00047			1000000000	00073	10000	00000	DEC	1827
•	00050			0400000000	00074	04000	00000	DEC	.5827
•	00051			737777777	00075	73777	77777	DEC	5827
•	00052			100000000	00076	10000	00000	DEC	1827
•	00053			737777777	00077	73777	77777	DEC	5827
•	00054			677777777	00100	67777	77777	DEC	-1827
•	00055			0400000000	00101	04000	00000	DEC	.5827
•	00056			0000000000	00102	00000	00000	DEC	0827
•	00057			7525252525	00103	75252	52525	DEC	33333333827
•	00060			040000000	00104	04000	00000	DEC	.5827
٠	00061			7652525252	00105	76525	25252	DEC	16666667B27
	00062	CAZ		0000000000	00106	00000	00000	DEC ERROLATION	0.0827 AZIMUTH INT
•	00063			7652525252	00107	76525	25252	DEC	16666667827
•	00064			000000000	00110	00000	00000	DEC	0827
•	00065			0125252525	00111	01252	52525	DEC	·16666667827
•	00066			0000000000	00112	00000	00000	DEC	0827
٠	00067			1000000000	00113	10000	00000	DEC	1827

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#### SPURT OUTPUT NO. 210 TEOSTE+06/19/64 INTER NOTES CARDS LI ID LABEL TA STATEMENT LOC F JKB Y 0400000000 00114 04000 00000 DEC ·5827 00070 737777777 00115 73777 77777 DEC -.5827 00071 1000000000 00072 00116 10000 00000 DEC 1827 00117 73777 77777 737777777 DEC -.5827 00073 00074 677777777 00120 67777 77777 -1827DEC 0400000000 00075 00121 04000 00000 .5827 DEC 00076 0000000000 00122 00000 00000 DEC 0827 00077 7525252525 00123 75252 52525 DEC -.33333333827 00100 0400000000 00124 04000 00000 DEC ·5827 00101 7652525252 00125 76525 25252 DEC -.16666667B27 00102 CDOP 0000000000 00126 00000 00000 DEC 0.0827 DOPPLER INT ERPOLATION COEFFICIE 00103 7652525252 00127 76525 25252 DEC -.16666667827 00104 0000000000 00130 00000 00000 DEC **OB27** 00105 0125252525 00131 01252 52525 DEC .16666667827 00106 0000000000 00132 00000 0000C DEC 0827 00107 1000000000 00133 10000 00000 DEC 1827 00110 0400000000 00134 04000 00000 DEC .5827 737777777 00111 00135 73777 77777 DEC -.5B27 00112 1000000000 00136 10000 00000 DEC 1827 00113 737777777 00137 73777 77777 DEC -.5827 677777777 00114 00140 67777 77777 DEC -182700115 0400000000 00141 04000 00000 DEC .5827 00116 0000000000 00142 00000 00000 DEC 0827 00117 7525252525 00143 75252 52525 DEC -.3333333B27 00120 0400000000 00144 04000 00000 DEC .5B27 00121 7652525252 00145 76525 25252 DEC -.16666667827 00122 INITSUB ENTRY 00146 61000 0000C LOCATION OF OUTPUT TABLE IN LI

	• • • •	SPURT	OUTPUT NO. 210 TEOSTE*06/19/64		••••
CARDS	L1 ID LABEL	TA STATEMENT	LOC F JK	B Y NOTES	
				413	
	001.23	STR A+I (R4)	00147 1501	A)	
•	00124	ENT 85+500	00147 1301	0 00104	
	00125 R1	ENT 86+2	00151 1260	0 00002	
	001 26	ENT A+JUNK+3	00152 1100	0 00203	
	001 27 R2	STR A+L(R3)	00153 1501	0 00157	
	00130	SUB A+3	00154 2100	0 00003	
	00131	STR A+L(R3+3)	00155 1501	0 00162	
	00132	ENT B7+2	00156 1270	0 00002	
•	00133 R3	ENT Q+W(000)	00157 1003	0 00000 SET T	O JUNK+3
•	00134	MUL W(Y)	00160 2203	000060	
•	00135	RSH AQ+27D	00161 0300	0 00033	
•	00136	STR A*L(R4) ENT B5*50D ENT B6*2 ENT A*JUNK+3 STR A*L(R3) SUB A*3 STR A*L(R3+3) ENT B7*2 ENT Q*W(000) MUL W(Y) RSH AQ*27D ADD Q*W(B7)	00162 2603	7 00000 SET T	O JUNK
	00137	BJP B7*R3+1 STR Q*W(B6) ENT A*4 ADD A*L(R3) BJP B6*R2 ENT Q*3 RPL Y+Q*L(R4) ENT A*W(Y) ADD A*W(DELY) STR A*W(Y) BJP B5*R1 EXIT	00163 7270	00 00160 ONE V	ALUE OF A COMPUTED
	00140 R4	STR Q#W(B6)	00164 1403	6 00000 SET T	O OUTPUT TABLE
•	00141	ENT A+4	00165 1100	00 00004	
•	OC142	ADD A+L(R3)	00166 2001	.0 00157	
•	00143	BJP B6+R2	00167 7260	0 00153 THREE	VALUES OF A COMPUTED
•	00144	ENT Q+3	00170 1000	00 00003	
	00145	RPL Y+Q+L(R4)	00171 3401	0 00164	
•	00146	ENT A+W(Y)	00172 1103	O OOOGC COMPU	TE NEW VALUE OF Y
•	00147	ADD A+W(DELY)	00173 2003	0 00177	
•	00150	STR A+W(Y)	00174 1503	0 00060	LUCK OF Y UKEN
•	00151	8JP 85*K1	00176 (250	0 00177 21 AV	FOE2 OF A OPEN
•	00152 00153 DELY	EXII	00177 001	U UU140	O2027 INCOEMENT O
٠	OUISS DELY	0012172702	00177 0012	F Y	OZBZI INCREMENI U
	00154 JUNK	RESERVE 12D	00200 0000	000000	
	00155 AZ	EQUALS ACQAZIM			
•	00156 EL	EQUALS ACQELEV			
•	00157 F	EQUALS JUNK+8D			
•	OG160 FIRST	EQUALS JUNK+1			
•	00161 SECOND	EQUALS JUNK+2			
•	00162 AAZ	RESERVE 153D	00214 0000	000000	
•	UU163 AEL	RESERVE 153D	00445 0000	00000	
•	OOLE TATEOR	KESEKAE 1530	01127 4100	000000	
•	OCL CTARTAZ	ENT AND AND S	01127 0100	0 42074 471411	TH LIMITS
•	00165 STARTAZ	ID TECTS	01130 [[03	0 630/4 AZIMU	In CINIIS
	00170	SIIR AHH (A7MAY) HADOS	01131 0100	0 01311	
•	00170	IP FNDA7+1	01132 2103	0 01332	
	00172	ENT A+W(A7MAY)	01134 1103	0 011332	
	OD173 ENDAZ	STR A+W(AZ+3)	01135 1503	0 63074	
	00174	ENT A+W(A7+3)	01136 1103	0 63074 OVERS	HOOT CORRECTION
	00175	SUB A+W(AZ+2)+APOS	01137 2163	0 63073	
	00176	STR A+A	01140 1504	0 00000	
•	00177	SUB A+W(BOUNCE)+ANEG	01141 2173	0 0131C	
	OC200 ENDAZBOUN	C JP TEST3	01142 6100	0 01315	
	00201 STARTEL	ENT A+W(EL+3)+APOS	01143 1163	0 63100 ELEVA	TION LIMITS
•	00202	JP TEST5	01144 6100	0 01322	
	00203	SUB A+W(ELMAX)+ANEG	01145 2173	0 01330	
•	00204	ENT A+W(ELMAX)+SKIP	01146 1113	0 01330	
*	00205	RESERVE 12D EQUALS ACQAZIM EQUALS ACQELEV EQUALS JUNK+8D EQUALS JUNK+1 EQUALS JUNK+1 EQUALS JUNK+1 EQUALS JUNK+2 RESERVE 153D RESERVE 153D RESERVE 153D RESERVE 153D ENTRY ENT A*W(AZ+3)*APOS JP TEST2 SUB A*W(AZMAX)*APOS JP ENDAZ+1 ENT A*W(AZ+3) STR A*W(AZ+3) STR A*W(AZ+3) STR A*W(AZ+3) STR A*W(AZ+3) SUB A*W(AZ+3) SUB A*W(AZ+3) SUB A*W(BOUNCE)*ANEG JP TEST3 ENT A*W(EL+3)*APOS JP TEST5 SUB A*W(ELMAX)*ANEG ENT A*W(ELMAX)*SKIP ENT A*W(EL+3)	01147 1103	0 63100	

## INTER SPURT OUTPUT NO. 210 TEOSTE\*06/19/64

CARDS	L1 ID LABEL	TA STAT	EMENT	LOC	F JKB Y	STORE DIFFERENCE OF AZO  STORE ROUNDED AZI  MOVE AZZ  STORE DIFF. AZZ  MOVE AZ3  STORE DIFF. AZ3  SPECIFY AZ A TABLE SPECIFY AZIMUTH BUFFER  STORE DIFF OF ELO
	00206 ENDEL	STR	A+W(EL+3)	01150	15030 63100	
ē	00207	SUB	A=W(EL+2)+APOS	01151	21630 63077	
	OC210	STR	A*A	01152	15040 0000C	
	00211	SUB	A+W(BOUNCE)+ANEG	01153	21730 01310	
•	00212	JP	TEST1	01154	61000 01303	
•	00213	ENT	A+W(AZ)	01155	11030 63071	
•	00214	SUB	A+W(AZ+1)	01156	21030 63072	
•	00215	STR	A+W(F)	01157	15030 00210	STORE DIFFERENCE OF AZO
•	00216	ENT	A+W(AZ+1)	01160	11030 63072	
	00217	STR	A+W(AZ)+APUS	01161	15630 63071	
•	00331	SUB	A*200*5KIP	01162	21100 00200	
•	00221	STD	A#200	01166	15030 00213	STORE BOUNDED A71
•	00222	ENT	AAU(A7A2)	01165	11030 63073	STOKE KOUNDED AZI
•	00223	STR	A-W(AZ+2)	01166	15030 63073	MOVE A72
	00225	SUR	A+W(A7)	01167	21030 63072	HOVE AZZ
	00226	STR	A+W(F+1)	01170	15030 00211	STORE DIFE. A72
	00227	ENT	A+W(AZ+3)	01171	11030 63074	STORE STITE ALL
	00230	STR	A+W(AZ+2)	01172	15030 63073	MOVE AZ3
•	00231	SUB	A+W(AZ)	01173	21030 63071	
	00232	STR	A#W(F+2)	01174	15030 00212	STORE DIFF. AZ3
•	00233	ENT	A+400	01175	11000 00400	
	00234	STR	A-W(LAP)	01176	15030 01326	
•	00235	ENT	84*AAZ	01177	12400 00214	SPECIFY AZ A TABLE
•	00236	ENT	B5+L(AZ [MADD)	01200	12510 63442	SPECIFY AZIMUTH BUFFER
•	00237	RJP	INTSUBR	01201	65000 01336	
•	00240	ENT	A+W(EL)	01202	11030 63075	
•	00241	208	A-W(EL+1)	01203	21030 63076	CTO-5 DISS OF SIG
	00242	ENT	A-W(FLA1)	01204	11030 00210	STURE DIFF OF ELU
•	00243	STR	A-W(EL ) - APOS	01205	15630 63076	
_	002 44	3115	A*W(EL)*APOS  A*200*SKIP  A*200 A*W(F+3) A*W(EL+2) A*W(EL+1) A*W(EL+1) A*W(EL+1) A*W(EL+3) A*W(EL+2) A*W(EL+2) A*W(EL+2) A*W(EL+2) A*W(EL+2) B5**L(ELEVADD) INTSUBR A*W(R+1) A*W(R+1) A*W(R+1) A*W(R+1) A*W(R+1) A*W(R+2) A*1 AQ*1 B4**L(RANGEADD) A*W(B4) A*W(R+2)	01200	15050 03075	
•	00245	SUB	A*200*SKIP	01207	21100 00200	
	00246 00247	ADD	A+200	01210	20000 00200	
•	00247	STR	A+W(F+3)	01211		STORE ROUNDED EL1
•	00250 00251	ENT	A+W(EL+2)	01212	11030 63077	Name of the second
•	00251	STR	A+W(EL+1)	01213	15030 63076	MOVE EL2
•	00252 00253	508	A+W(EL)	01214	21030 63075	STORE DIEE SLA
•	00254	SIK	AWW(F+I)	01215		STORE DIFF EL2
•	00254	STP	AAW(ELYS)	01210	11030 63100 15030 63077	MOVE EL 3
	00255 00256	SUR	A+W(FL)	01211	21030 63075	HOVE EES
	00257	STR	A+W(F+2)	01220		STORE DIFF EL3
	00260	ENT	B4+AFI	01222		SPECIFY EL A TABLE
	00261	ENT	B5+L(ELEVADD)	01223		SPECIFY ELEVATION BUFFER
	00262	RJP	INTSUBR	01224	65000 01336	
•	00262 00263	ENT	A+W(R+1)	01225	11030 01461	
	00264 00265	STR	A+W(R)	01226	15030 01460	
•	00265	ADD	A+W(R+2)	01227	20030 01462	
•	00266	ADD	A+1	01230	20000 00001	
•	00267	RSH	AQ#1	01231	03000 00001	
	00270	ENT	B4*L(RANGEADD)	01232	12410 63445	Avenue aluae in messa
•	00271	STR	A+W(B4)	01233		AVERAGE RANGE IN INTERRANGE
•	00272	ENI	A*HIK+Z)	U1234	11030 01462	

	SPURT OUTPUT NO. 210
INTER	TEOSTE * 06/19/64

CARDS	L1 ID LABEL			LOC	F JKB Y	
	00273	STR	A+W(p+1)	01235	15030 01461	
	00274	ENT	A+W(CRANGE)	01236	11030 63057	
	00275	STR	A+W(R+2)	01237	15030 01462	R VALUES MOVED
	00276	ENT	A+W(RDOT)	01240	11030 01463	
•	00277	SUB	A+W(000T+1)	01241	21030 01464	
	003.00	STR	A+W(F)	01242	15030 00210	STORE DIFFERENCE OF ROOTO
	00301	ENT	Q+W(RDOT+1)	01243	10030 01464	STORE STITLE STITLE
	00302	STR	O+W(RDOT)	01244	14030 01463	MOVE ROOTO
•	00303	ADD	O#W(BIAS)	01245	26030 01335	
•	00304	STR	A*W(R+1) A*W(CRANGE) A*W(R+2) A*W(RDOT) A*W(RDOT+1) A*W(F) Q*W(RDOT+1) Q*W(RDOT) Q*W(RDOT) Q*W(BIAS) Q*W(F+3)	01246	14030 00213	STORE ROUNDED AND BLASED DOP1
	00205	ENT	A*W(RDOT+2) A*W(RDOT+1) A*W(RDOT) A*W(F+1) Q*W(RANGEDOT) W(RECC) AQ*30D Q*H(UKS) W(RANGEDOT) AQ*24D W(CONVERSION) AQ*26D Q*W(RDOT+2) Q*W(RDOT) Q*W(F+2) B4*E(RANGEADD) Q*W(F+2) B4*E(RANGEADD) Q*W(B4) W(RANGEUNIT) AQ*10 W(F+1) AQ*28D Q*W(F+3) Q*W(F+3) W(LAP)	01247	11030 01445	
•	00306	STR	A+W(pDOT+1)	01250	15030 01465	MOVE ROOTS
•	00307	SIIR	A-W(RDOT)	01250	21030 01463	HOVE ROOTE
•	00310	STP	A-W(E-1)	01251	15030 01403	STORE DIE POOTS
•	00310	ENT	O-W(PANGEDOT)	01252	10030 63042	STORE DITE ROOTE
	00311	MIII	M(DECC)	01255	22030 00057	
•	00312	B S H	A0 + 3(ID	01254	03000 00031	R24
•	00313	SIIR	UPPLINC J	01256	27030 00056	024
•	00314	Mult	W(PANGEDOT)	01257	22030 62062	
•	00316	PSH	An+24n	01251	03000 03002	R24
•	00317	MIII	MICUNIESCIUM)	01261	22030 00050	524
•	00327	PSH	A0+260	01261	03000 00001	DODA BR CPS
•	00320	STR	044(PDOT+2)	01262	14030 01465	MOVE ROOTS
	00321	SIIR	0=W(000T)	01264	27030 01463	MOVE ROOFS
	00322	STR	0+W(E+2)	01265	14030 01703	STOP DIE, ROOTS
	00323	ENT	RANI (RANGEADD)	01266	12410 63445	STOR DITT NOOTS
•	00324	ENT	0+W(84)	01267	10034 00000	
•	00326	MIII	W(pancelinit)	01270	22030 01334	
	00327	H S H	A0+10	01271	03000 00010	TIME 828
	00330	MIII	W(F+1)	01272	22030 00010	Tane geo
•	00330	RSH	AQ+28D	01273	03000 00034	RANGE CORRECTION 88
•	00332	ADD	0+W(E+3)	01274	26030 00213	
•	UC333	STR	Q+W(F+3)	01275	14030 00213	ADD TRANSIT TIME DOPPLER CHANG
						E
•	00334	CL	W(LAP)	01276	16030 01326	COCCLEY CODD, CO. A. TABLE
	00335	ENT	84+ADUP	01277	12400 00676	SPECIFY DUPPLER A TABLE
•	00336	ENT	B5+L(DUPPADD)	01300	12510 63444	SPECIFY DUPPLER BUFFER
•	00337	RJP	INISUBK	01301	65000 01336	
•	00340	EXII		01302	61010 01127	
•	UU341 TEST1	ENI	A+W(EL+3)	01303	11030 63100	
•	00342	STR	A+W(EL+2)	01304	15030 63077	
•	00343	STR	A+W(EL+1)	01305	15030 63076	
•	00344	STR	A*W(tL)	01306	15030 63075	
•	00344 80005	JP	ENUEL+5	01307	61000 01155	056 0147937 04
•	UU346 BUUNCE	0010	W(LAP) B4*ADDP B5*L(DOPPADD) INTSUBR  A*W(EL+3) A*W(EL+2) A*W(EL+1) A*W(EL) ENDEL+5 431634	01310	UUIU4 31634	DEC •0167B27 OV ERSHOOT CONSTANT
_	00347 TEST2	SUB	A • W ( AZMIN ) • ANEG ENDAZ + 1 A • W ( AZMIN ) ENDAZ A • W ( AZ + 3 ) A • W ( AZ + 2 )	01311	21730 01331	
	00350	JP	ENDAZ+1	01312	61000 01136	
	00351	ENT	A+W(AZMIN)	01313	11030 01331	
	00352	JP	ENDAZ	01314	61000 01135	
	00353 TEST3	ENT	A+W(AZ+3)	01315	11030 63074	
	00354	STR	A+W(AZ+2)	01316	15030 63073	

### SPURT OUTPUT NO. 210 INTER

TEOSTE + 06/19/64

CARDS	L1 ID	LABEL	TA	STAT	EMENT	LOC	F JKB		NOTES	
	00355			STR	A*W(AZ+1) A*W(AZ) ENDAZBOUNC+1 A*W(ELMIN)*ANEG EDEL-1 A*W(ELMIN) ENDEL	01317	15030	63072		
•	00356				A=W(AZ)	01320	15030	63071		
•	00357			JP	A+W(AZ) ENDAZBOUNC+1	01321	61000	01143		
•		TEST5		SUB	A+W(ELMIN)+ANEG	01322	21730	01327		
•	00361			JP	ENDEL-1	01323	61000	01147		
*	00362			JP	A+W(ELMIN)	01324	11030	01327		
1	00364			0	ENDEL	01325	01000	01130		
•		ELMIN		_	341217	01327			DEC005827	
•	00366	ELMAX		02000	000000	01330	02000	00000	DEC .25B27	
•	00367	AZMIN		7534	121727	01331	75341	21727	DEC32B27 MI NIMUM ALLOWABLE AZIM	
•	00370	AZMAX		1243	656050	01332	12436	56050		
•	00371	REVOLUTION		10000	000000	01333	10000	00000	DEC 1827	
•	00372	RANGEUNIT		00000	006553	01334	00000	06553	DEC .00000005836 .1 USEC./2SEC.	
•	00373	BIAS		1334	330200	01335	13343	30200	DEC 750000.588 DOPPLER BIAS AND ROUNDING	
•	00374	INTSUBR		ENTR					BUFFER LOC IN B5, A LOC IN B4	
	00375			ENT	Q+W(F)	01337	10030	00210	FFIRST = A-1F-1+A1F1+A2F2 FIRST POINT COMPUTED SET UP LOOP	
•	00376			MUL	W(B4)	01340	22034	00000		
•	00377				AQ+27D	01341	03000	00033		
•	00400				Q+W(FIRST)	01342	14030	00201		
•	00401				Q+W(F+1) W(1+B4)	01343	10030	00211		
•	00402				AQ+27D	01344	02000	00001		
	00404				Q+W(FIRST)	01346	26030	00201		
	00405				Q+W(FIRST)	01347	14030	00201		
•	00406				Q+W(F+2)	01350	10030	00212		
•	00407				W(2+B4)	01351	22034	00002		
•	00410				AQ=27D	01352	03000	00033		
•	00411				Q+W(FIRST)	01353	26030	00201	FFIRST = A-1F-1+A1F1+A2F2	
•	00412 00413				Q+W(FIRST) B6+49D	01354	12600	00201	CET UP LOOP	
•	00414	R5		ENT	Q+W(F)	01356	10030	00001	JET OF LOUP	
	00415				W(3+B4)	01357	22034	00003		
	00416				AQ+27D	01360	03000	00033		
•	00417			STR	Q+W(SECOND)	01361	14030	00202		
**	00420				Q+W(F+1)	01362	10030	00211		
•,	00421				W(4+B4)	01363	22034	00004		
•	00422				AQ+270	01364	03000	00033		
•	00423				Q+W(SECOND)	01365	26030	00202		
•	00424				Q+W(SECOND) Q+W(F+2)	01366	19030	00202		
•	00425				W(5+B4)	01357	22034	00212		
	00427				AQ+27D	01370	03000	00003		
	00430			ADD	Q+W(SECOND)	01372	26030	00202	FSEC = A-1F-1+A1F1+A2F2	
•	00431				Q+W(SECOND)				SECOND POINT COMPUTED	

## SPURT OUTPUT NO. 210 INTER TEOSTE+06/19/64

		INTER	IED21E+00/TA/04
CARDS	L1 ID LABEL	TA STATEMENT	LOC F JKB Y NOTES
•	00432	SUB Q+W(FIRST)	01374 27030 00201
	00433	MUL W(TENTH)	01375 22030 01457
•	00434	RSH AQ = 32D	01376 C3000 0004C
•	00435	STR Q+W(JUNK)	01377 14030 00200 DIFFERENCE BETWEEN 4POINT INT.
			POINTS
•	00436	ENT A+W(F+3)	01400 11030 00213
•	00437	ADD A=W(FIRST)+APOS	01401 20630 00201
•	00440	ADD A+W(LAP)	01402 20030 01326
•	OC441	RSH AQ#8D	01403 03000 00010
	00442	STR A+H(B5)	01404 15035 0000C FO LINEARLY INTERPOLATED
	OC443	LSH AQ+8D	01405 C7000 0001C
	00444	ADD A*#(JUNK) RSH AQ*8D	01401 20030 00201 01402 20030 01326 01403 03000 00010 01404 15035 0000C FO LINEARLY INTERPOLATED 01405 07000 00010 01406 20030 00200
	00445	RSH AQ#8D	01407
•	06446	STR A+W(1+B5)	01410 15035 00001 F1 LIN INT
•	UC447	LSH AQ+8D	01411 C7000 0001C
•	00450	ADD A+W(JUNK)	01412 20030 00200
•	00451	RSH AQ+8D	01413 03000 0001C
	OC452	STR A+W(2+B5)	01414 15035 00002 F2 LIN INT
	UC453	LSH AQ+8D	01406 20030 00200 01407 C3000 00010 01410 15035 00001 F1 LIN INT 01411 C7000 0001C 01412 20030 00200 01413 03000 0001C 01414 15035 00002 F2 LIN INT 01415 07000 0001C 01416 20030 00200
•	00454	ADD A+W(JUNK)	01416 20030 00200
•	00455	RSH AQ+8D	01417 03000 00010 01420 15035 00003 F3 LIN INT
•	DC456	STR A+W(3+B5)	01420 15035 00003 F3 LIN INT
•	00457	LSH AQ+8D	01421 07000 0001C
•	00460	STR A+W(3+B5) LSH AQ+BD ADD A+W(JUNK)	01422 20030 00200
•	00461	K2H AQ+8D	01423 03000 00010
•	00462	STR A+W(4+B5)	01421
	00463	LSH AQ+8D	01425 07000 00010
•	00464	ADD A+W(JUNK)	01426 20030 00200
•	00465	RSH AQ+8D	01427 03000 00010
•	00466	STR A+W(5+B5)	01430 15055 00005 F5 LIN INI
•	00467 00470	LSH AQ*8D ADD A*W(JUNK)	01432 20030 00200
•	00471	RSH AQ#8D	01432 20030 00200
•	00472	STR A+W(6+B5)	01433
•	00473	LSH AQ+8D	01435 07000 00010
•	U0474	ADD A+W(JUNK)	
•	00475	DCII AO-OD	01/27 02000 00010
	U0476	STR A+4(7+R5)	01440 15035 00007 F7 LIN INT
	00477	STR A*W(7+B5) LSH AQ*8D	01441 07000 00010 01442 20030 00200 01443 03000 00010 01444 15035 0001C F8 LIN INT
	00500	ADD A+W(JUNK)	01442 20030 00200
	00501	DCH VU* DD	01443 03000 00010
	00502	STR A+W(10+85)	01444 15035 0001C F8 LIN INT
	00503	I CH AOARD	01445 07000 00010
	00504	ADD A+W(JUNK)	01446 20030 00200
	00505	RSH AQ#8D	01447 03000 00010
	00506	STR A+W(11+B5)	01450 15035 00011 TEN VALUES LINEARLY INTERPOLAT
			ED
•	00507	ENT 84+3+84	01451 12404 00003
•	00510	ENT 85+10D+85	01452 12505 00012
•	00511	MOVE 1 * SECOND * FIRST	01453 10030 00202
		V 15: 2. W2	01454 14030 00201
	00512	8JP 86*R5	01455 72600 01356
•	00513	EXIT	01456 61010 01336

		INTER	SPURT DUTPUT NO. 210 . TEOSTE+06/19/64	• • • • • • • • • • • • • • • • • • • •
CARDS	L1 ID LABEL	TA STATEMENT	roc	F JKB Y NOTES
•	00514 TENTH 00515 R 00516 RDOT 00517	3146314631 RESERVE 3* RESERVE 3* RESERVE 1	01457 01460 01463 01466	31463 14631 C0000 00000 00000 00000 00000 00000

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## 13. ABSTRACT

The Haystack antenna is pointed by means of a Univac 490 Computer. In the Pointing System program is incorporated an interpolation routines to provide pointing information 250 times a second. This report describes the interpolation routine.

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